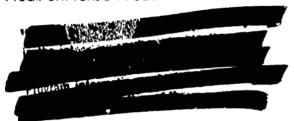
### NOTICE

THIS DOCUMENT HAS BEEN REPRODUCED FROM MICROFICHE. ALTHOUGH IT IS RECOGNIZED THAT CERTAIN PORTIONS ARE ILLEGIBLE, IT IS BEING RELEASED IN THE INTEREST OF MAKING AVAILABLE AS MUCH INFORMATION AS POSSIBLE



## Lyndon B. Johnson Space Center

Houston, Texas 77058



# PLAID EVALUATION OF VISUAL ACCESS FOR PDP BERTHING AND DEPLOYMENT

**APRIL 1980** 

(NASA-TM-81102) PLAID EVALUATION OF VISUAL ACCESS FOR PDP BERTHING AND DEPLOYMENT (NASA) 45 p HC A03/MF A01 CSCL 22A

N80-28410

Unclas G3/16 26415

# PLAID EVALUATION OF VISUAL ACCESS FOR PDP BERTHING AND DEPLOYMENT

Prepared by:

Approved by:

James L. Lewis, Head

Crew Station Design Section

Approved by: George C. Franklin, Acting Chief Spacecraft Design Division

#### PLAID EVALUATION OF VISUAL ACCESS FOR PDP BERTHING AND DEPLOYMENT

#### INTRODUCTION

The Plasma Diagnostics Package (PDP) is a small, pallet-mounted payload that must be deployed and berthed on-orbit by the Shuttle Remote Manipulator System (RMS). The PDP is berthed to a retention mechanism (REM) which is resting on an OSS pallet. The PDP is scheduled for STS-4; for this flight, the complement of payloads in the bay also includes an Induced Environment Contamination Monitor (IECM), which is also berthed to a REM, and a DFI pallet. Concerns about capabilities of the Payload Deployment and Retrieval System (PDRS)—namely the RMS--to perform PDP berthing and deployment culminated in a test request co-sponsored by the PDRSO and SPIDPO. The request is attached as Appendix A.

The test emphasized assessing the RMS operator's visual access (either by direct vision or CCTV) to the appropriate points of interest, as well as developing those marks or aids which would be required to complete berthing and deployment under different Orbiter lighting conditions. The original plan was to construct full-scale mockups of the OSS pallet, the PDP, and other hardware which could affect the visual access, then place these mockups in the one-g trainer to perform a preliminary assessment. A second phase of the plan was to place these mockups in the MDF and perform PDP berthing and deployment using the manipulator arm.

At the time this test request was introduced, no required pallet or experiment hard-ware was available, and the MDF--in the middle of an overall test to assess large payload deployment and retrieval--had a backlog of other tests waiting in line. As an alternative, the first phase of the PDP test was planned for completion using the Spacecraft Design Division's Design Performance Laboratory (DPL) computer system called PLAID.

#### PLAID DESCRIPTION

PLAID, which stands for Panel Layout Automated Interactive Design, is a 3-D computer graphics package which allows the user to construct models of real elements in three-space, assemble the elements as desired, then view the assembly from any specified eye point to any specified point on or around the assembly. Perspective can be enabled to produce a real-world picture of the scene.

#### TEST SETUP

For this application of PLAID, the PDP, pallet, Orbiter payload bay, the RMS arm, and other relevant elements were modeled in PLAID and assembled to represent the STS-4 configuration. Drawings supplied by the test sponsor were used to help ensure dimensionally accurate PDP and pallet/experiment models, and the PDRS data base used to ensure accurate placement of elements within the payload bay.

Several different assemblies were made, each one representing a "snapshot" of the PDP during the berth/deploy maneuvers. The cases selected for this study included the PDP as follows:

- berthed
- one foot up in Z
- two feet up in Z
- three feet up in Z

ten feet up in Z.

• six feet up in Z

Each of these six assemblies was then viewed from the six possible eye/camera points, including the RMS operagor's design eye point, the three payload bay bulkhead cameras (port forward and aft, starboard aft), and the two RMS-mounted cameras (elbow and wrist). For all viewpoints except the wrist camera, the look-to point was specified as the REM. For the wrist camera, the look-to point was along the camera's line-of-sight since this camera cannot pan and tilt. The fixed eye/camera locations, in Orbiter coordinates, are as follows:

VIEWPOINT	<u> </u>	<u> Ү</u> о	<u>z</u>	
Design eye point	-586.0*	-15.0	-464.2	
Forward port CCTV	-596.4	-71.5	-446.0	
Aft port CCTV	-1286.6	-87.0	-446.0	
Aft starboard CCTV	-1286.6	87.0	-446.0	

#### RESULTS

#### The Figures

Thirty-six views (i.e., six views of each of six assemblies) were produced during this test. Each view (Figures 1-36) has a numerical code which appears in the upper right corner of the figure. The numerical code can be interpreted as follows. The first digit (4 in all figures) indicates the STS-4 flight configuration. The next two digits indicate the PDP position above its berthed location (e.g., 00 shows PDP in its berthed position, 02 is with the PDP 2 feet above the berthed position, 06 is 6 feet up, etc.). The last digit indicates the view-from location as follows:

- Ol Operator's design eye point
- 02 Forward port CCTV
- 03 Aft starboard CCTV
- 04 Aft port CCTV
- 05 Elbow CCTV
- 06 Wrist CCTV

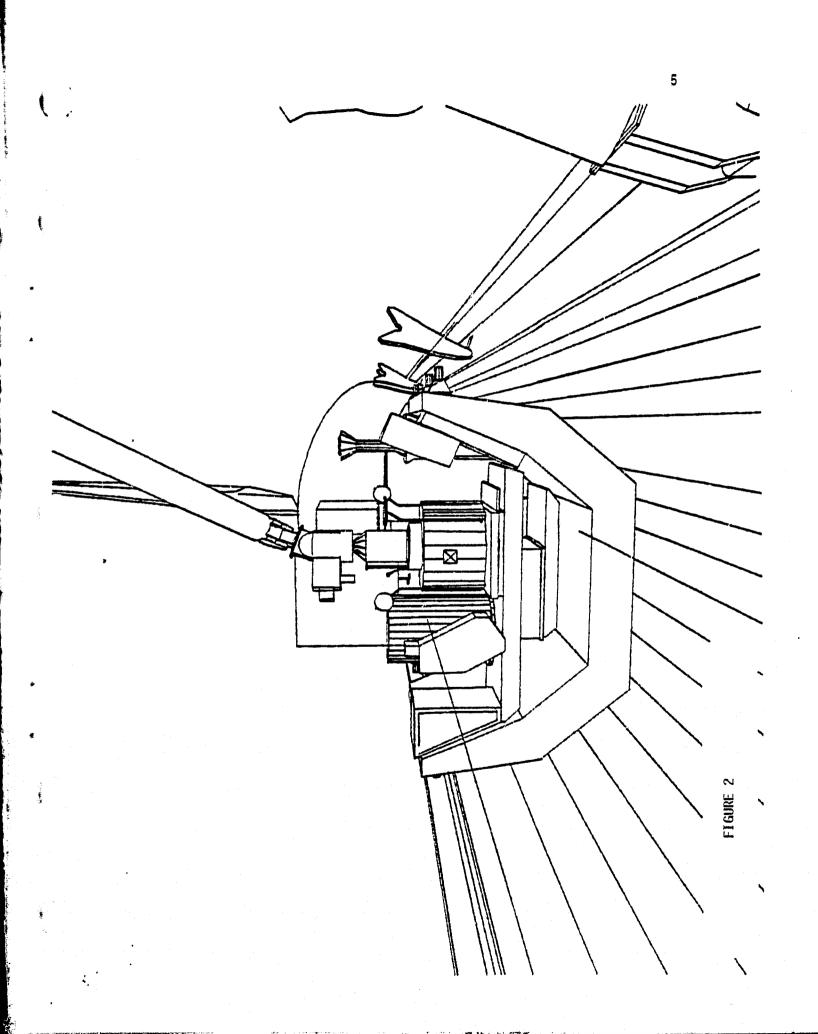
For example, 4103 is the view seen from the aft starboard CCTV camera during STS-4 of the PDP 10 feet above its berthed position.

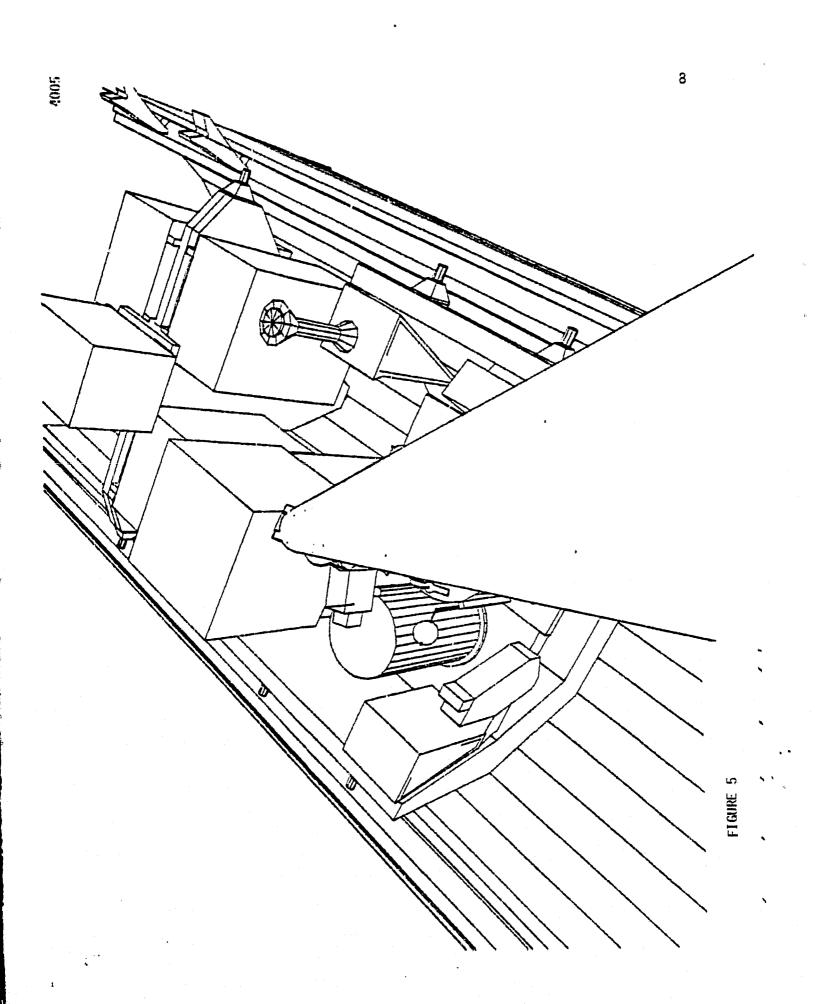
\*Set aft to view past forward bulkhead.

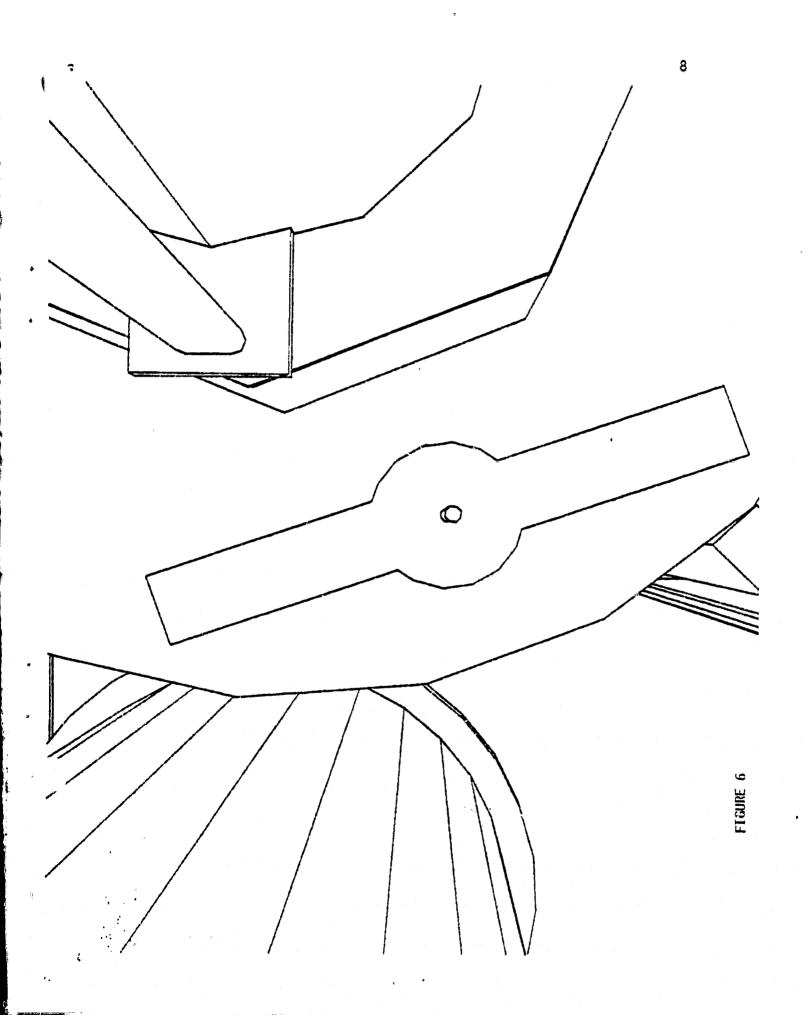
#### FIGURE IMPLEMENTATION

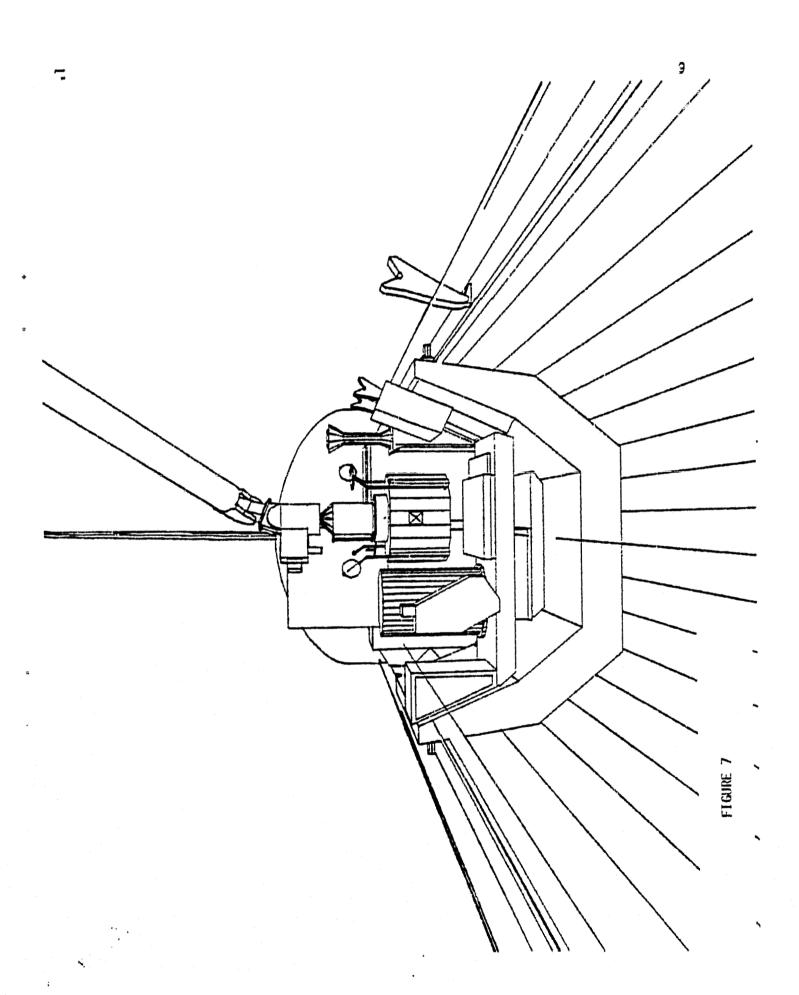
Views of the PDP from the two aft cameras are occluded by the DFI/IECM and the pallet thermal canister. The elbow camera views are blocked by the lower arm. However, this location may still be viable to provide visual access with the implementation of the elbow camera wedge. Additional views should be run with this wedge configuration. Forward visual access is good, either from the RMS operator's design eye point location or via the port forward CCTV camera. However, this visual access does not provide good cues for determining PDP X location. The wrist camera may be useful in providing X cues by sighting in the periphery of the views on the experiment adjacent (i.e., port of) the PDP, the Solar Flare X-Ray Polarimeter (see Figures 4006, 4016, 4026, 4036, 4066, and 4106). For example, a series of short lines at the approximate X station could provide a track down which the end effector TV would be flown.

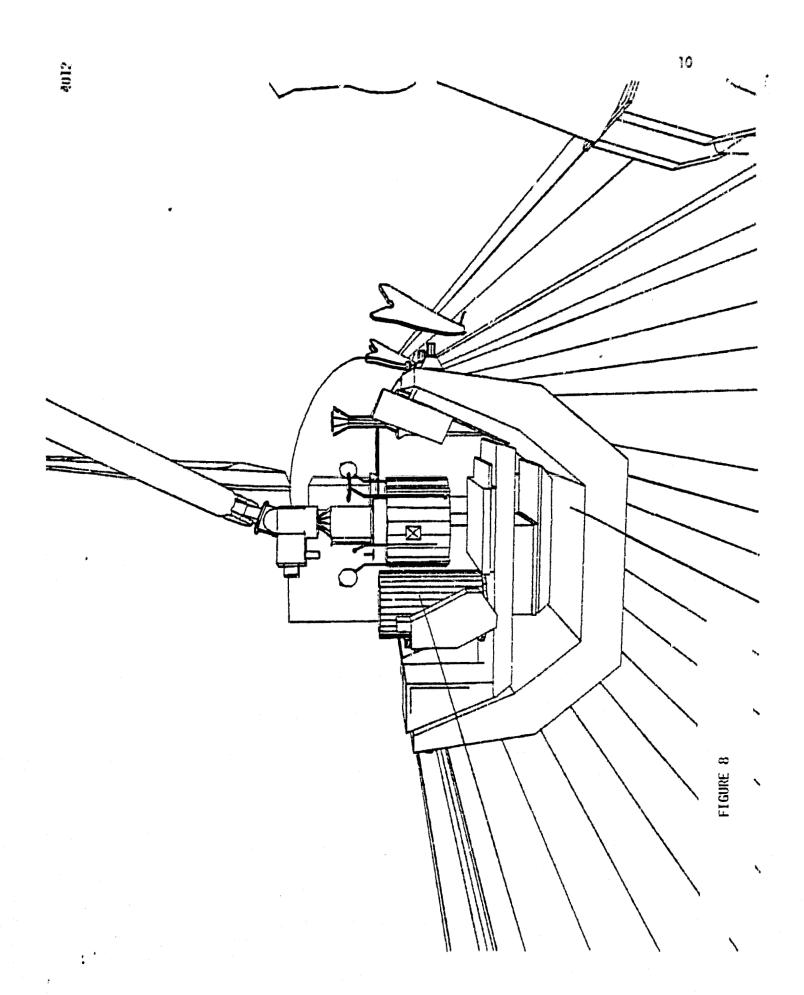
Further evaluations as specified in the test request are planned later this year in the MDF. The results of this PLAID study will be utilized in identifying preliminary marks for the test setup and in providing flight-configuration views for comparison with and extrapolation from the MDF configuration.

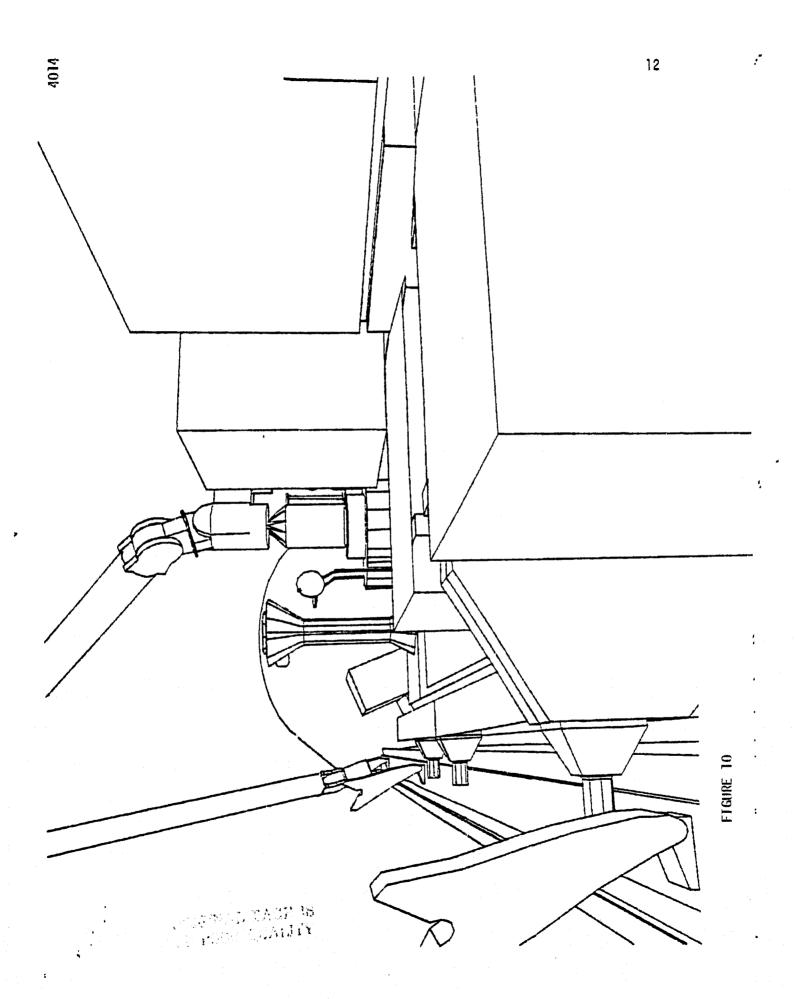


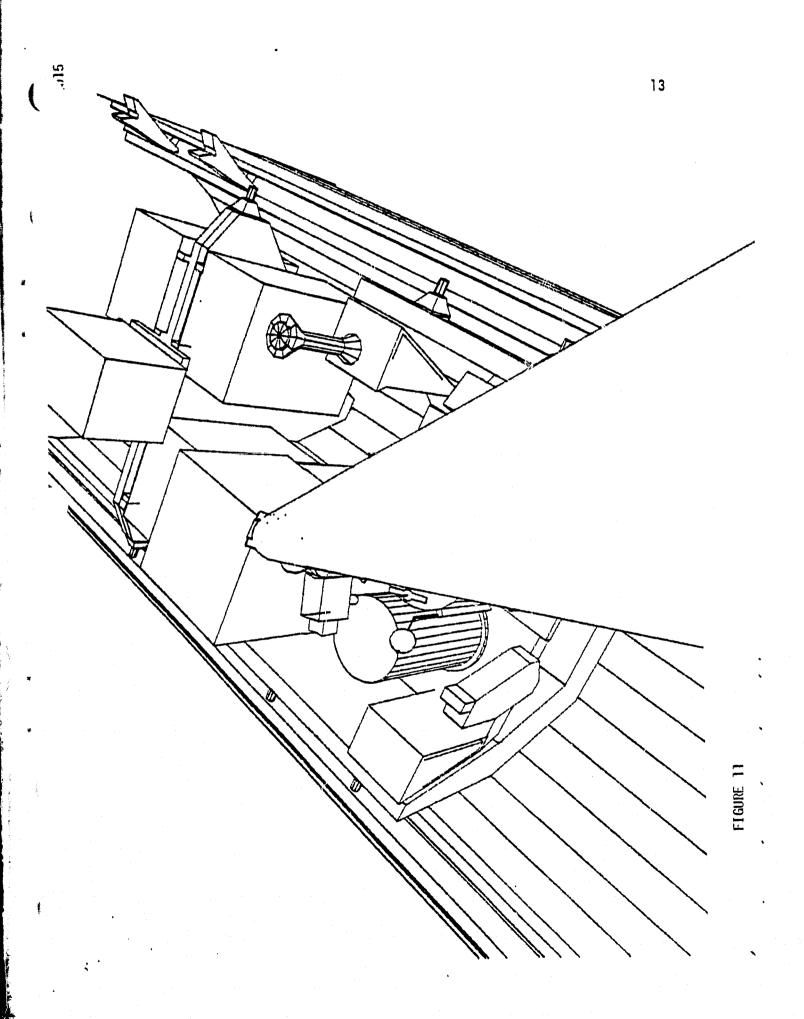


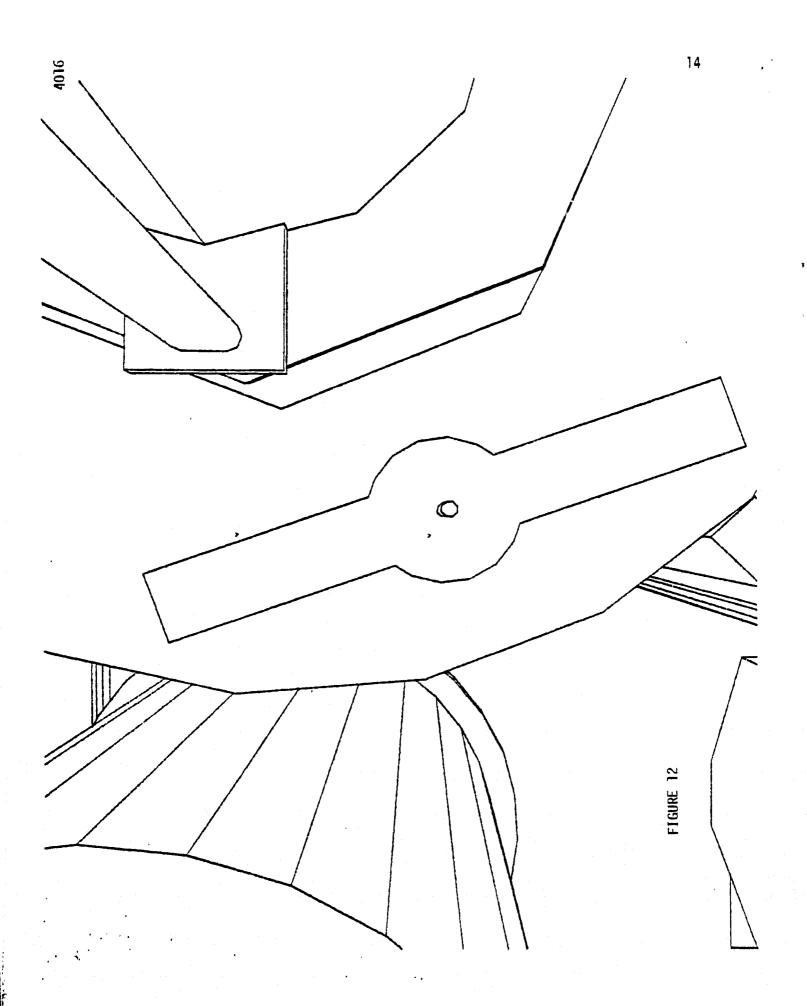


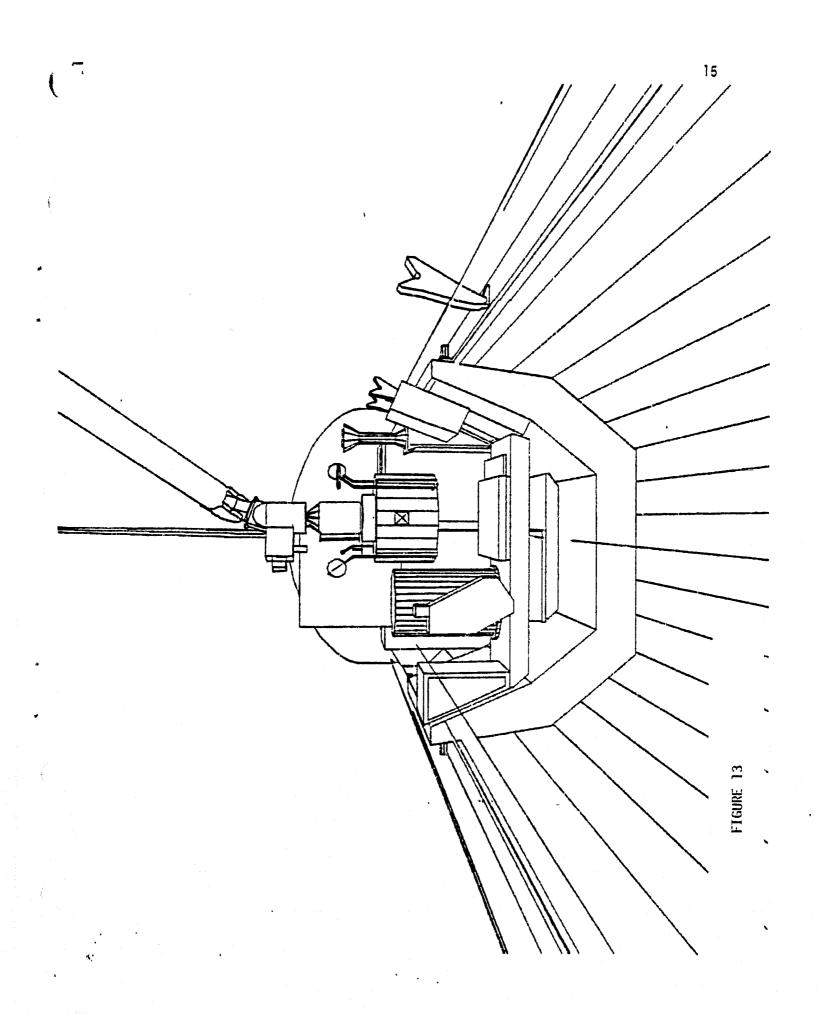


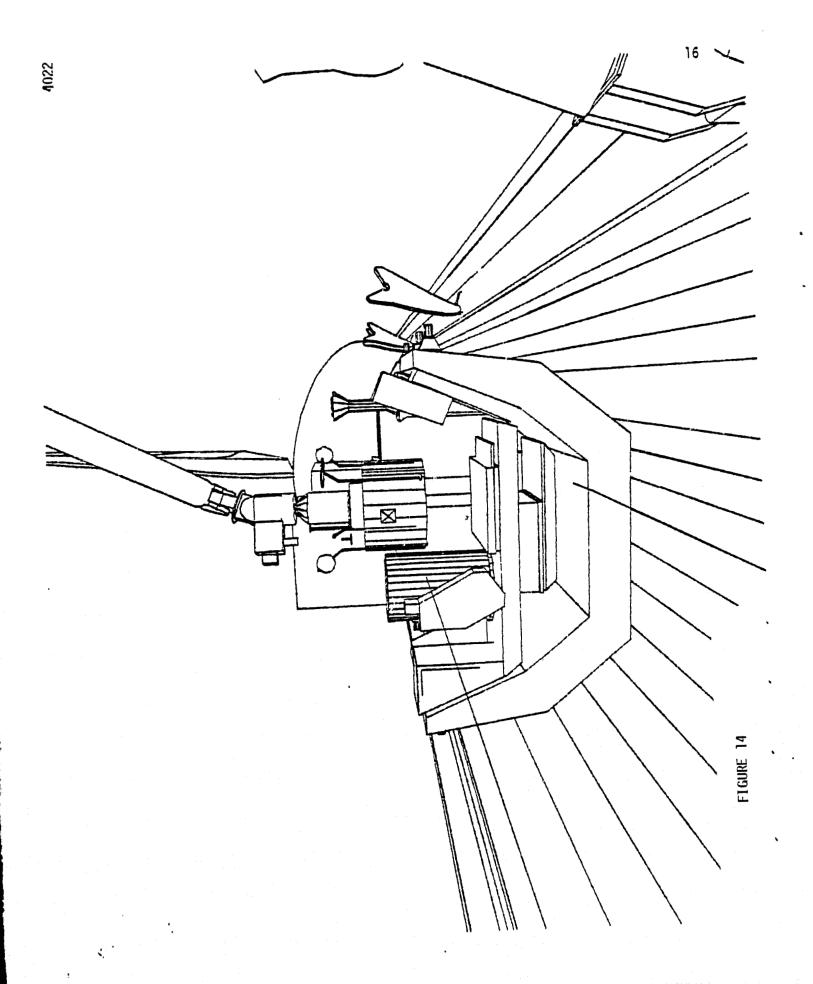


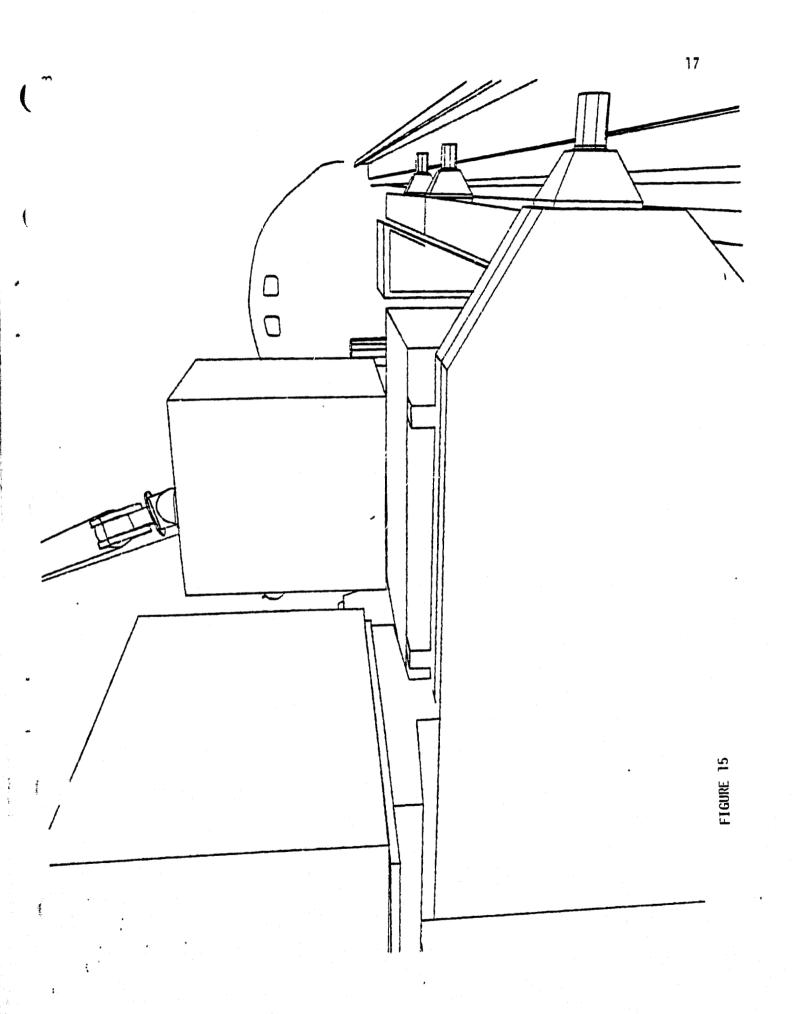


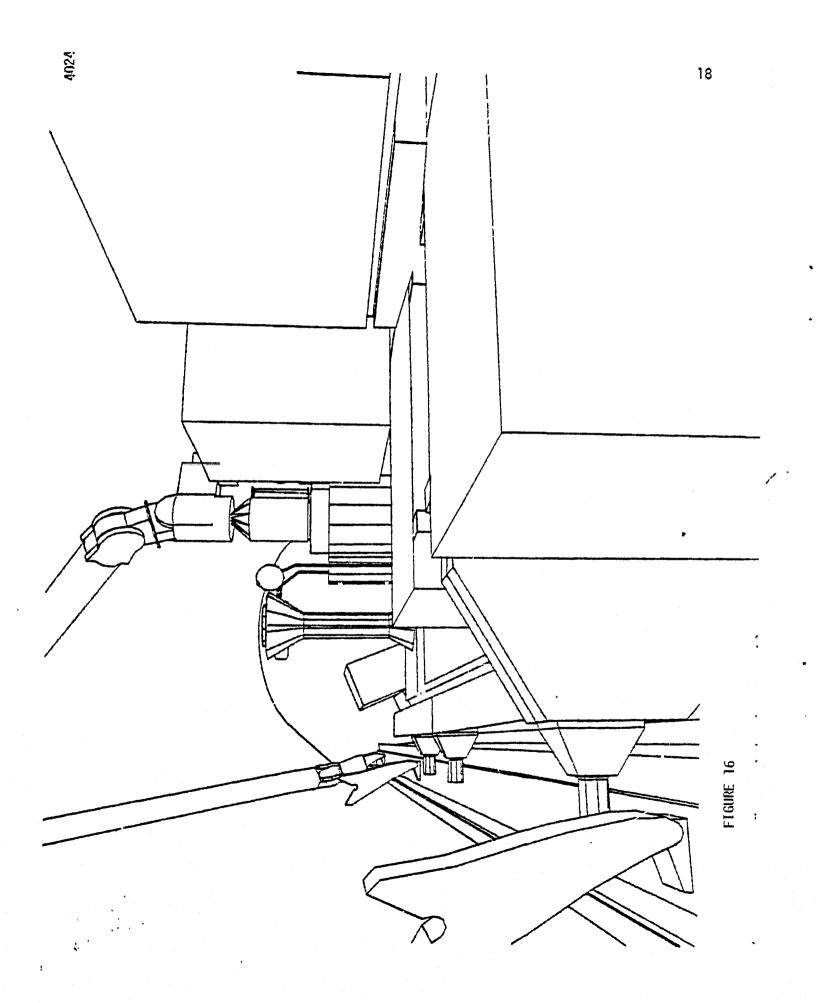


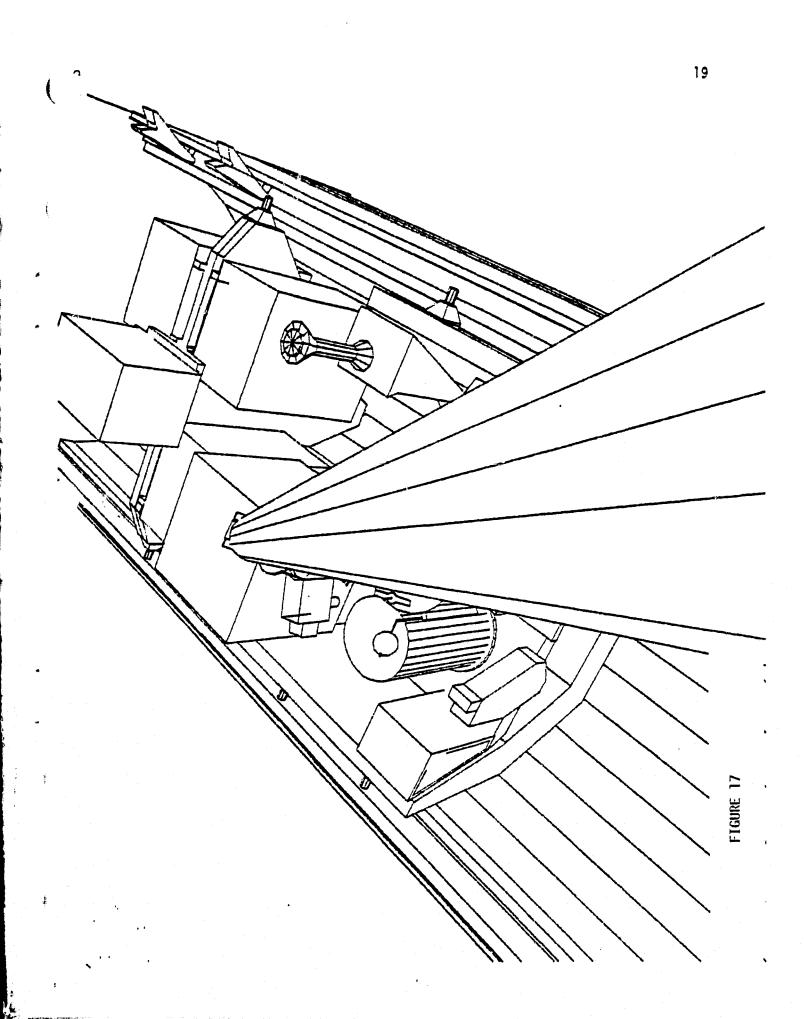


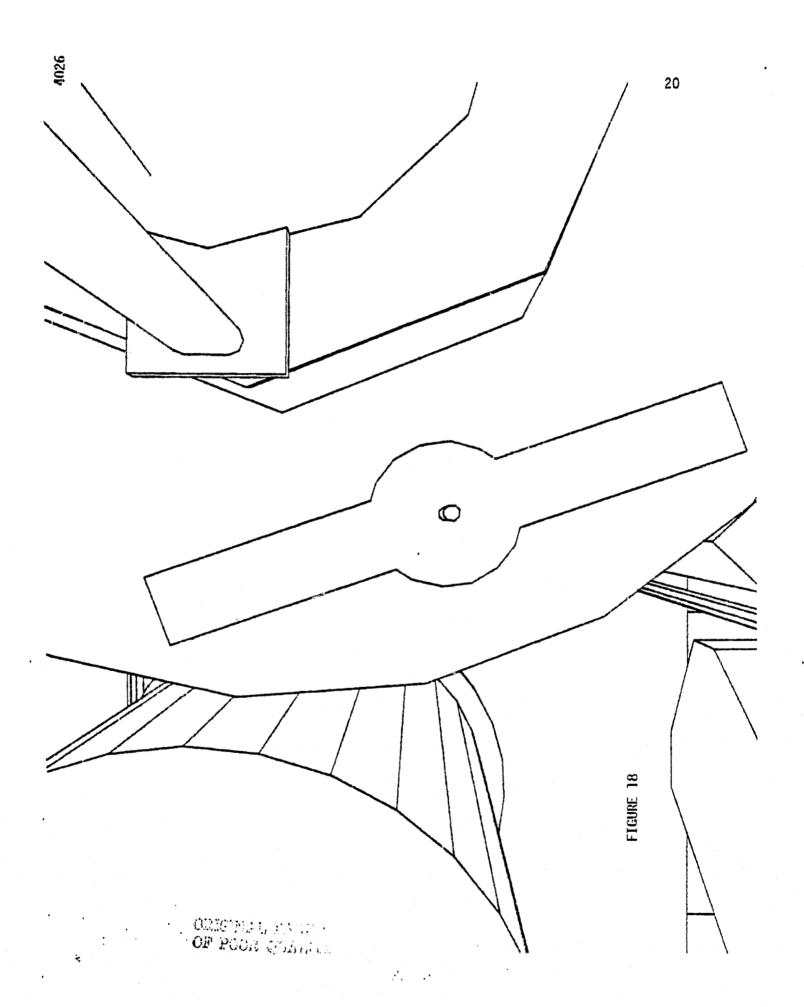


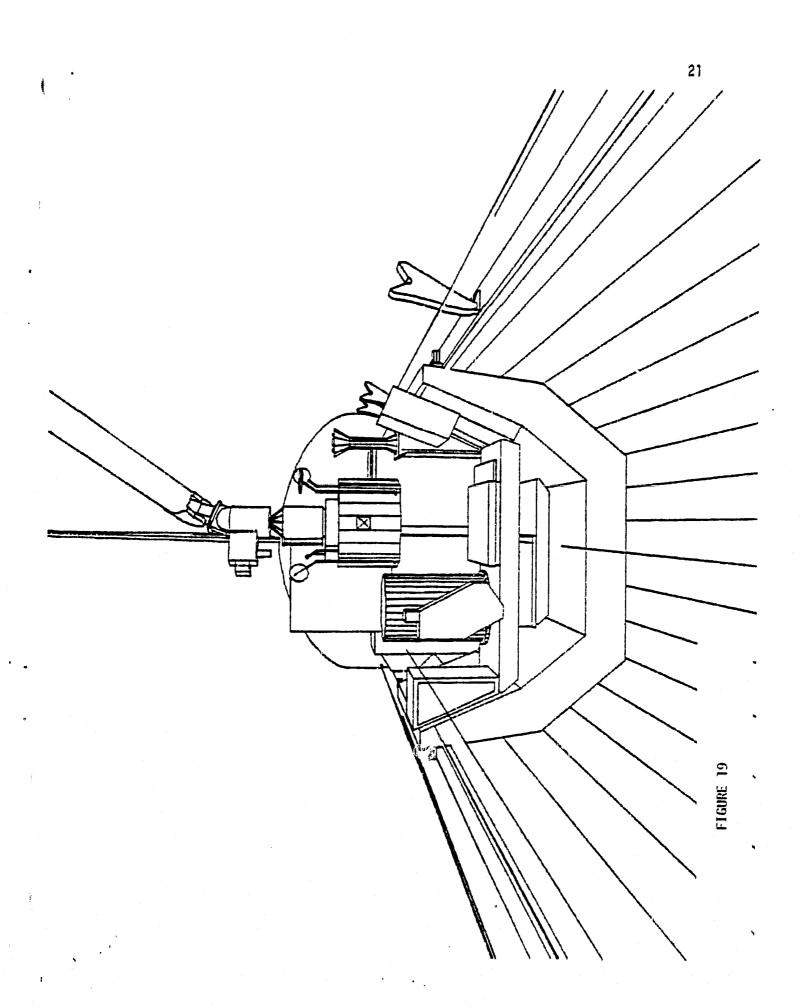


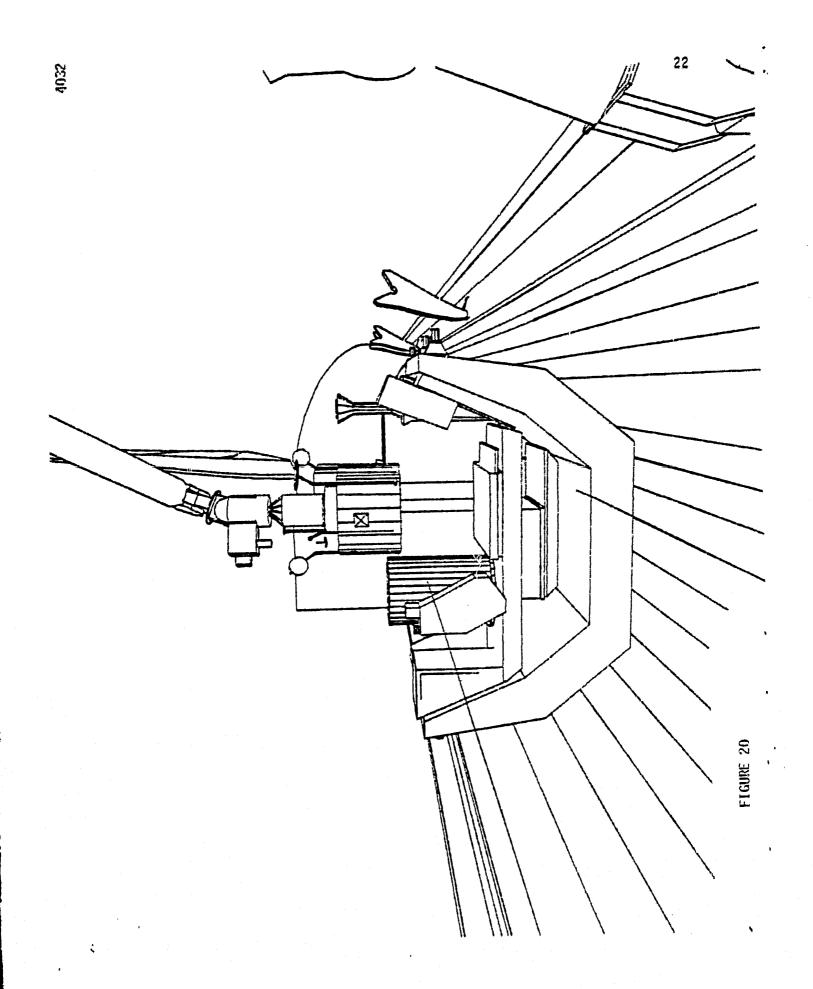


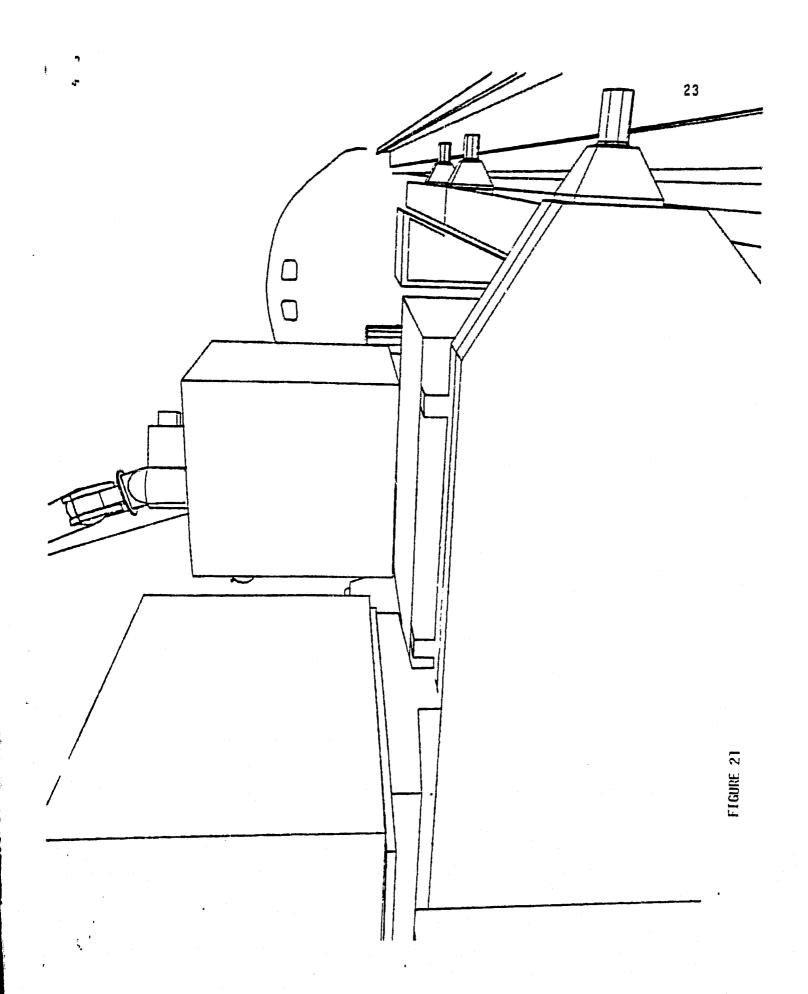


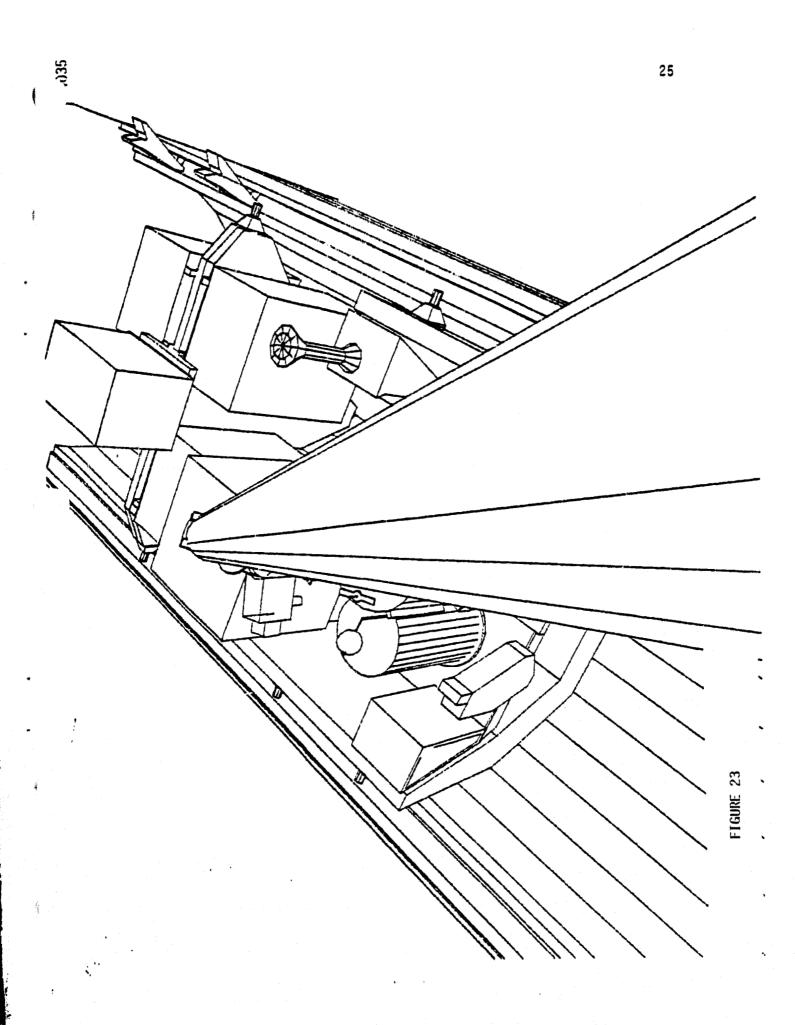


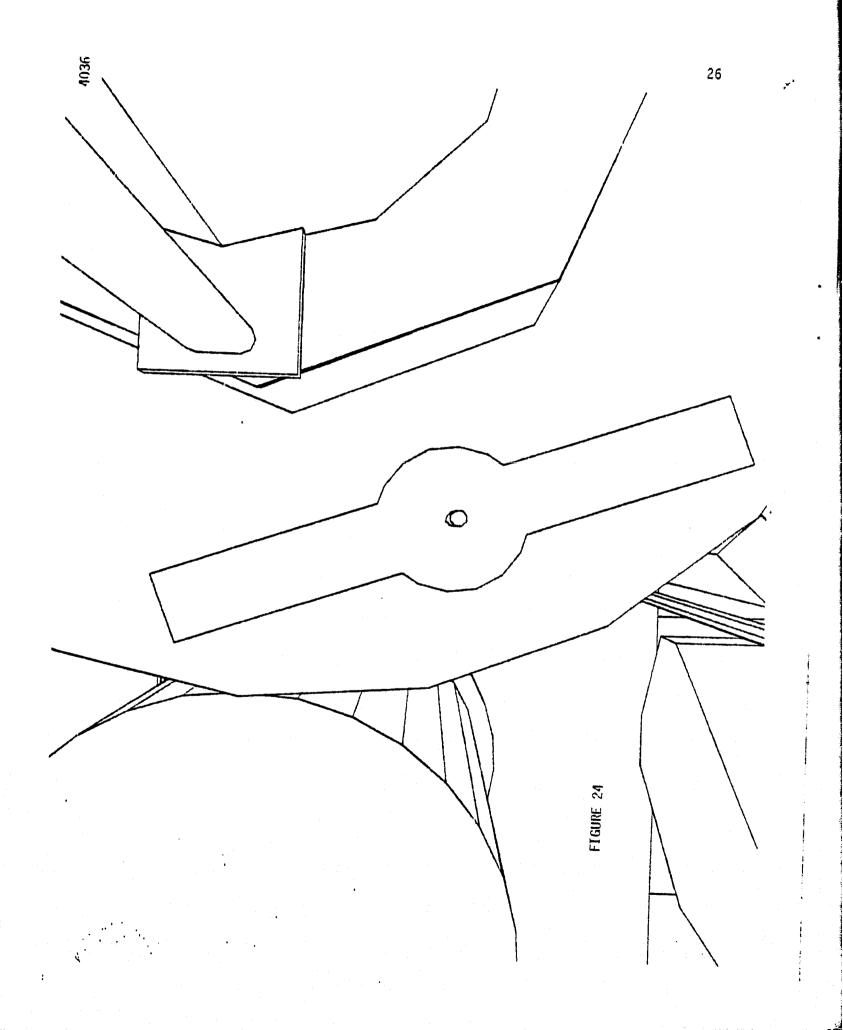


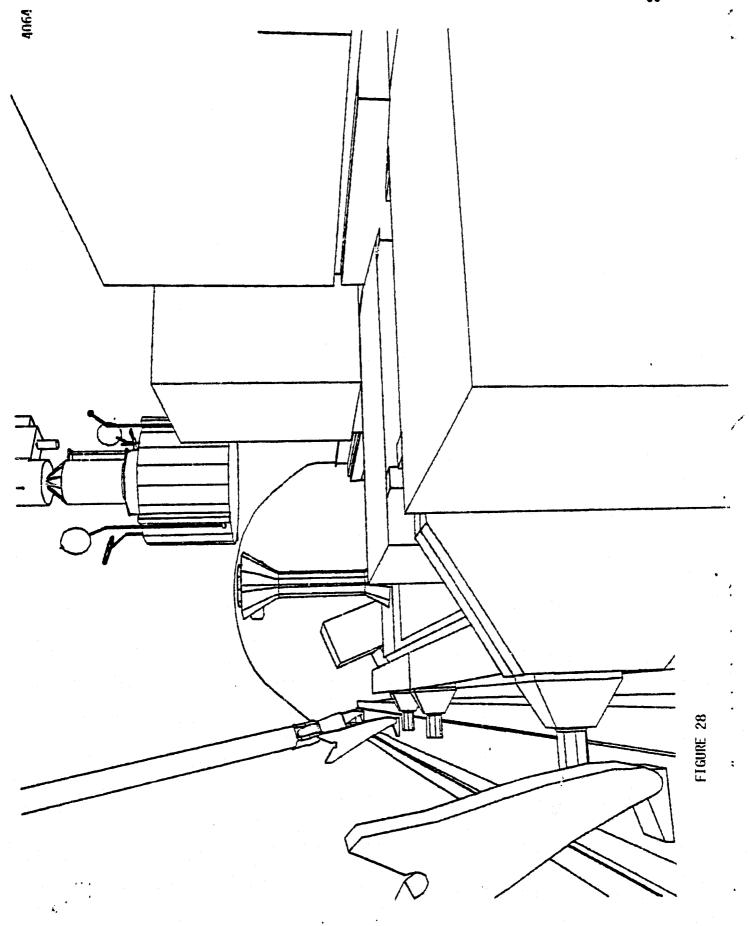


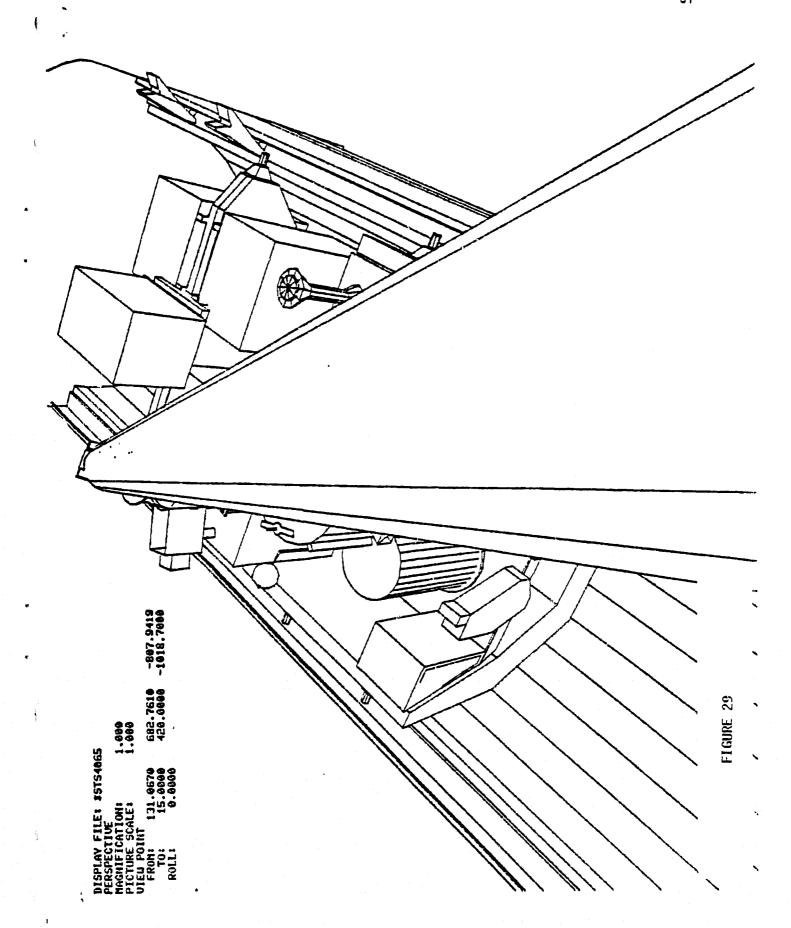


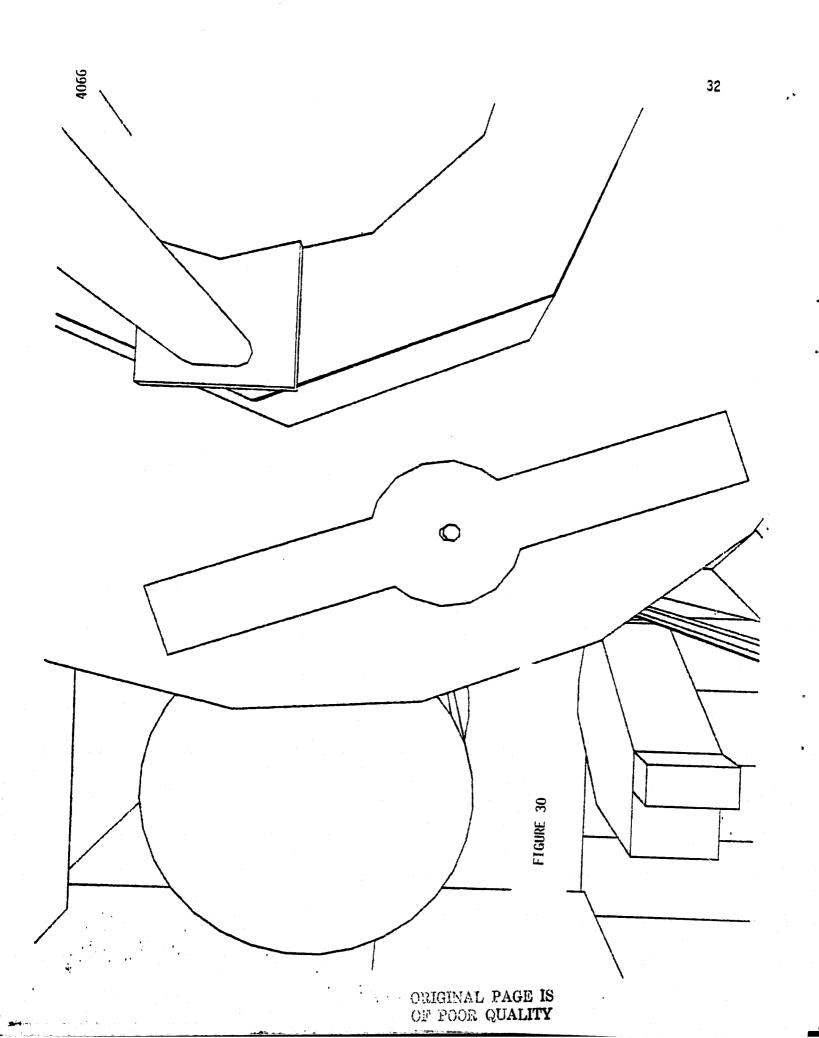


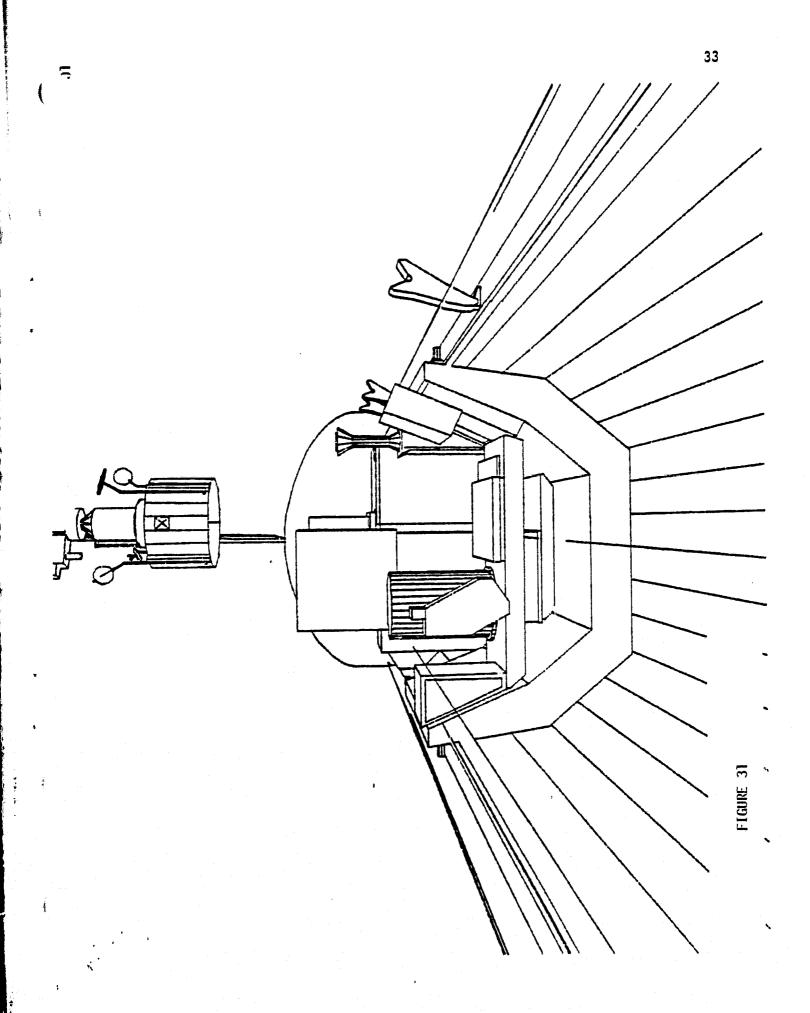


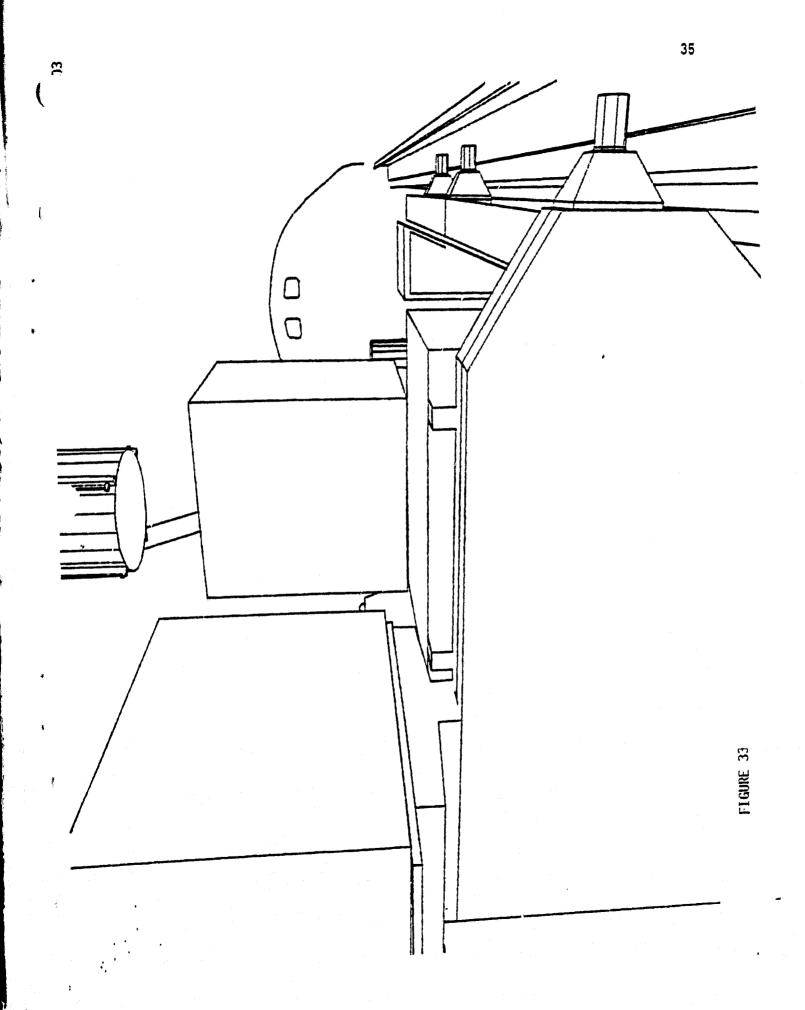


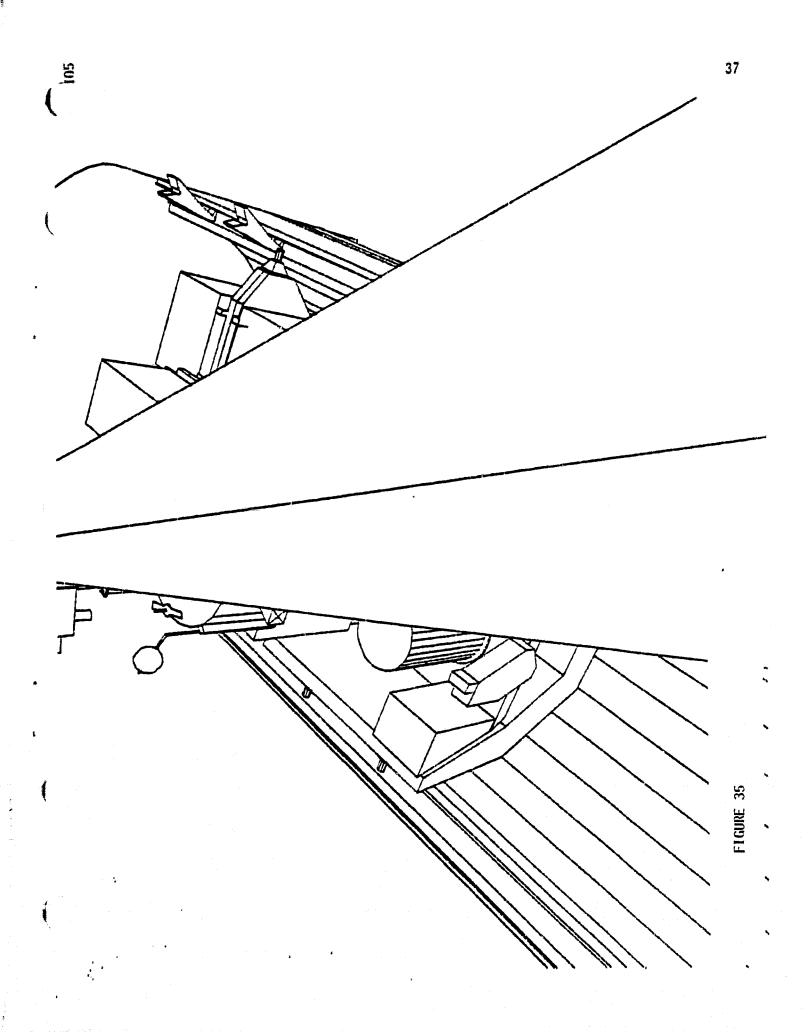


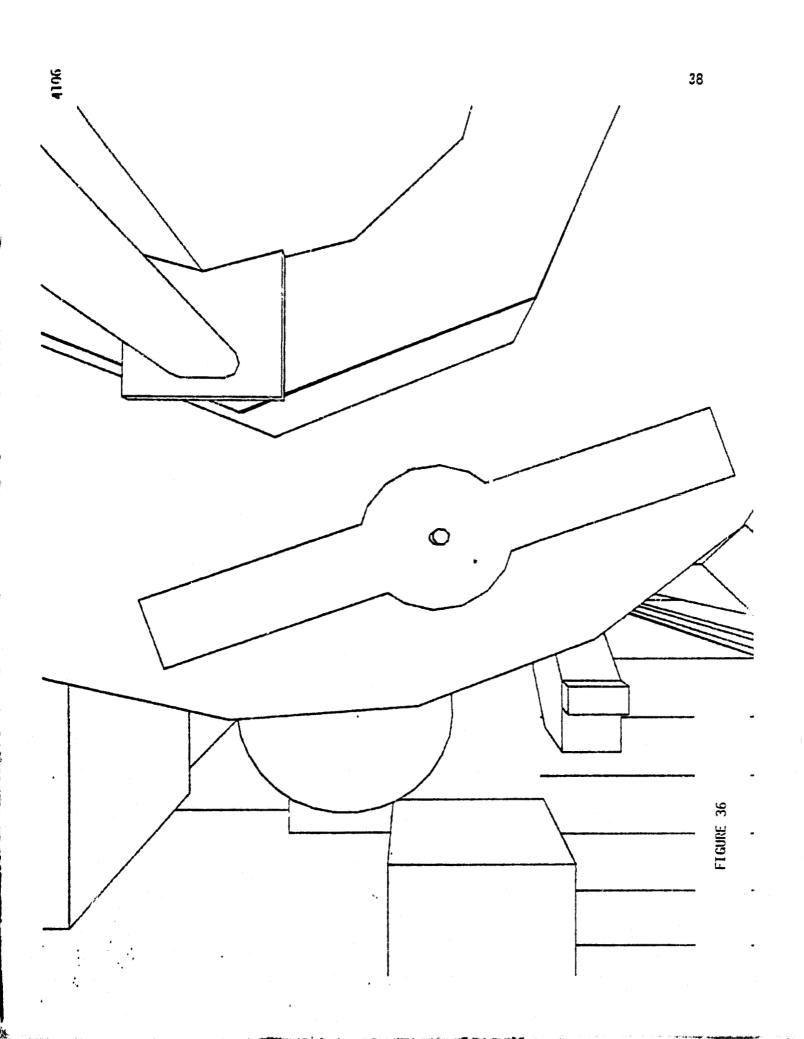












APPENDIX A: FACILITY UTILIZATION REQUEST NO. 175

Development of Visual Cues for Berthing the

Plasma Diagnostics Package (PDP)

			FOR SDD USAGE ONLY		
		DATE RESE VES			
FACILITY UTILIZATION	IN REQUEST	ACOUEST 50.			
		4690831 70'	1751	Page of	
IDENTIFYING INFORMATION					
CATE OF REQUEST	INITIATOR/EXTENSION		INTERATING O	FFICE	
Amust 00 1070	D laceble /C Da		PF		
April 20, 1979	P. Jaschke/S. Da	V15	*YPE		
	(Initial phase co August 15, 1979	uld be condu earlier)	cted s	UITED X UNSUITED	
-AD-LITIES  XX Manipulator Development Facili		_	uation lab		
E&D Mockup Facility	·' [	☐ Lighting Evaluation Lab ☐ Trainer Support			
Precision Air Bearing Floor		Water Immersion Facility			
☐ Design Performance Lab	ב	Other			
***LE:		<b>~·</b>	L	(200)	
Development of Visual Cues for Berthing the Plasma Diagnostics Package (PDP)					
CESCRIPTION. (Include purpose, test of	bjectives and Auidelines	)(Use additional	sheets, if no	ecessary,)	
The test objective is to eva					
berthing of the PDP from/to	the OSS pallet rel	ease mechanis	sm (REM),	assess the Orbiter/	
PDP configuration (including	cCTV) under vario	us lighting o	conditions	, and define any	
markings required for PDP be					
phase could be conducted in to the PDP berthing interface	the 1-y trainer, the and to permit an	early defini	ition of be	erthing markings	
required. The second phase s	should be conducted	in the MDF,	using the	same basic mockup	
hardware, but with man-in-th	ne-loop the simulat	ion of the Pi	OP berthin	g operation.	
REMARKS. (Special hardware comicemen	uta. Support equipment :	afety procedures	etc.)		
REMARKS, (Special hardware requirements, support equipment, safety procedures, etc.)					
Required Hardware (For detail see test plan)					
1. OSS pallet (or equivalent) and PDP mockup. The OSS pallet should include mock-up of instrumentation adjacent to the PDP to provide an accurate visual representation for the RMS operator.					
2. Good fidelity hardware of the REM berthing interface. It is desirable that this be available for use during both phases of this test.					
3. Complete payload bay configuration, including the DFI pallet.					
↑ Overhead crane as required.					
APPROVALS					
REQUESTER REQUEST!		FACILITY CIRECTO	9 50¢	REVIEW BOARD	
		•	1		

150 524- 1403 (135 75)

- 5. CCTV bulkhead cameras (2 aft and one forwart--port), RMS cameras.
- 6. Payload bay lights.
- 7. 1-g Orbiter cabin and payload bay mockup for Phase I, MDF Orbiter cabin and payload bay mockup for Phase II.

#### TEST PLAN

#### Introduction

This test is required to evaluate RMS operator visual accessibility for extraction and berthings of the Plasma Diagnostics Package (PDP) from the OSS pallet.

#### **Objective**

The major test objective is to assess visibility for the PDP handling and berthing operations and to define any markings or other visual aids required for successful RMS operations.

#### Test Description

The test will be conducted in a full-scale mockup of Orbiter cargo bay. To provide an early assessment of requirements for PDP berthing alignment markings or aids, the test could be conducted in the 1-g trainer. RMS operator visual accessibility to PDP handling and berthing could be evaluated through use of the overhead crane, to simulate vertical motion of the PDP to/from its berthing platform. The primary objective of this early phase of the test is to define required markings or aids proposed direct and indirect (CCTV) viewing, under various light environments. The second phase of the test will evaluate the markings or aids proposed from the first phase, but with man-in-the-loop simulations for RMS handling and berthing of the PDP to its baseline berthing platform. The PDP/OSS-1 pallet should be located in the cargo bay identical to the actual flight location for STS-5 (see attached figure). In each phase of the test, the cargo bay elements that effect the operator viewing capabilities should also be included, such as the DFI pallet.

#### Test Procedures

- 1. With the PDP in its stowed position, adjust the CCTV cameras pan and tilt units for optimium viewing of the berthing interfaces. Adjust the TV cameras zoom for optimium viewing.
- 2. Photograph the different CCTV views on a TV monitor. Also photograph the view out the aft port window.
- 3. Have the facility personnel move the PDP a foot forward on the REM. (away from the latch position).

- 4. Repeat step 2.
- 5. Have the facility personnel move the PDP a foot toward port on the REM, (away from the latch position).
- 6. Repeat step 2.
- 7. With overhead crane, lift PDP about a foot above the REM.
- 8. Repeat step 2.
- 9. With overhead crane, raise the PDP to about six feet above the REM. Note the out the window and TV views. Lower the PDP on to the REM.
- 10. If the camera views were not adequate to deploy and both the PDP, find new locations for the cameras where they will do the job. Follow the above procedures to verify the views from the new camera locations.
- 11. Place berthing markings on the PDP and REM as required to facilitate the deploy/berthing task. When the optimium markings are determined, the photographs as in step 2 and also closeup documented photos of the markings.

This sequence is written for phase one, the second phase is similar, but with the RMS and operator performing the PDP motions and berthing.

#### Data Requirements

The documentary photographs indicated by the test procedures and the subjective comments of the operators will be the only required data.

#### Test Configuration

The cargo bay should be configured with the OSS pallet/PDP located as shown in the attached figure. The DFI Pallet/IECM mockup should be installed to represent a realistic visual situation. The OSS pallet and associated experiments mockup should be of sufficient fidelity to provide to the RMS operator the expected viewing for berthing and translating the PDP to its latching interface. A flight-like PDP release mechanism (REM) should be used for at least the man-in-loop phase of the test. The PDP should be dimensionally correct and include the grapple fixture located as indicated by drawings which will be provided. The baseline CCTV configuration should be provided, including the elbow camera. If the 1-g trainer is used for the initial test phase, facilities should be provided to position a camera representative of the elbow camera location when berthing conditions deploying the PDP. Variable lighting conditions should be provided during the tests.